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SIMULATION OF  
AXISYMMETRIC  
SOURCE-SINK  
FLOWS IN A  
ROTATING FLUID  
ANNULUS.

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## PURPOSE OF WORK

- ① Validate Nuclear Electric Ltd CFD code FEAT (Finite-Element Analysis Toolbox) against rotating fluid flows\*
- ② Explore some interesting physics.

→ Look for simplest possible rotating fluid flow hence :-  
2D - axisymmetric flow  
Isothermal - no buoyancy

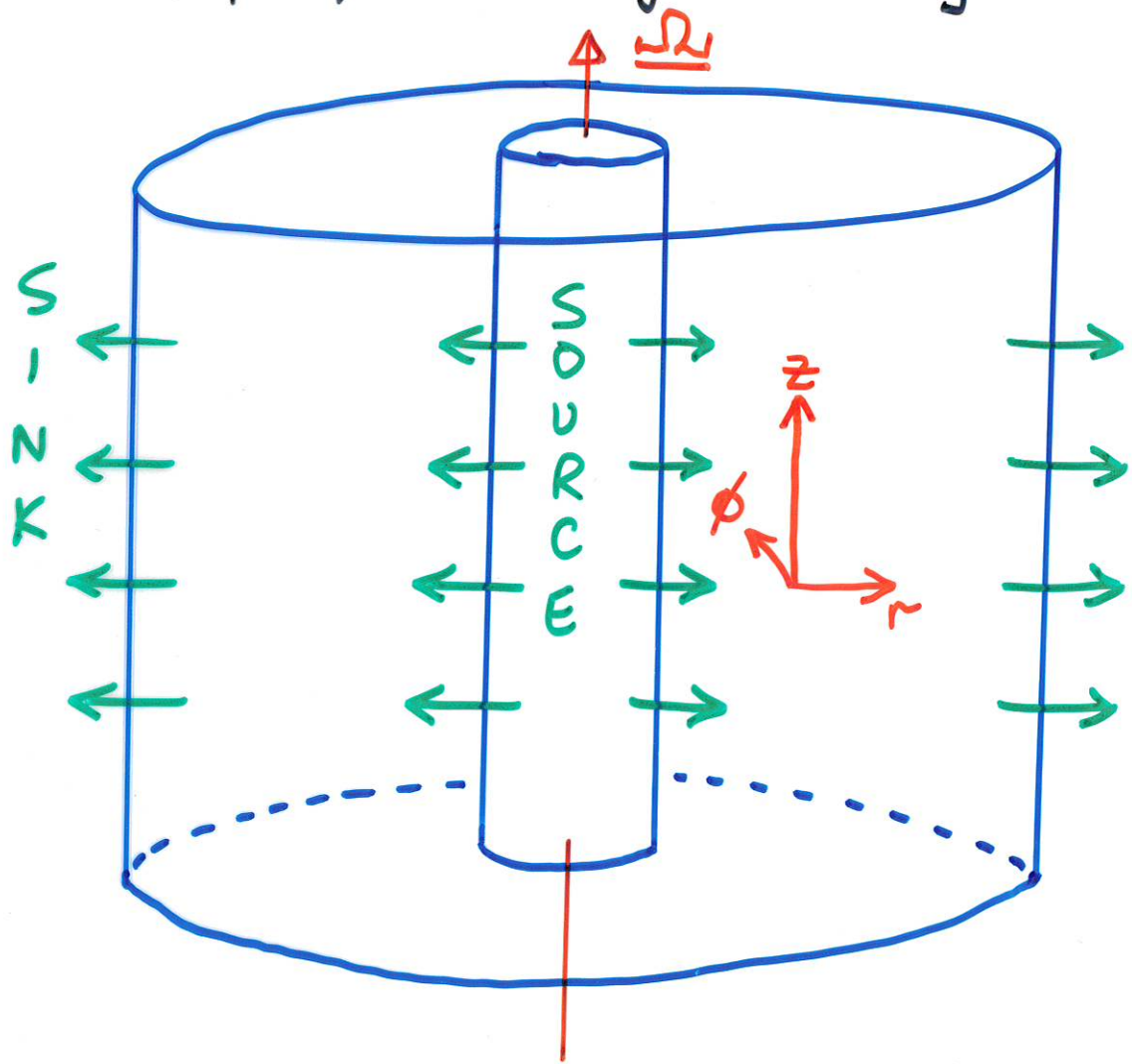
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\* ... and for me learn some CFD!

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THE SYSTEM :

- Rotating fluid annulus
- Fluid pumped through radially

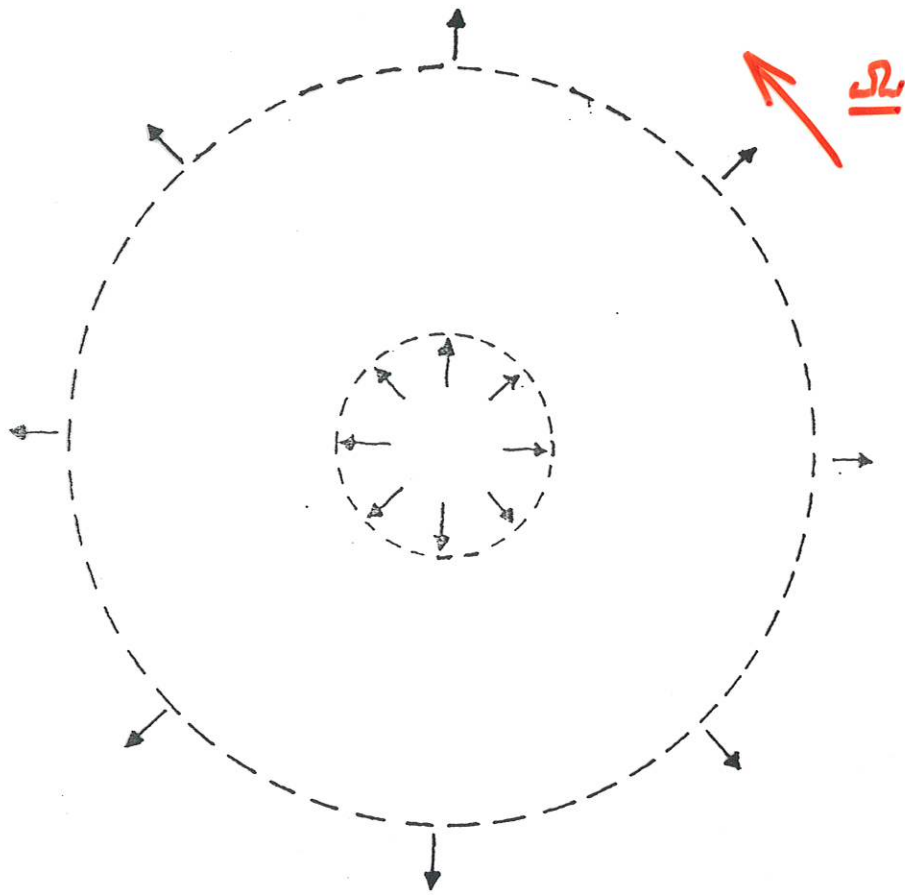


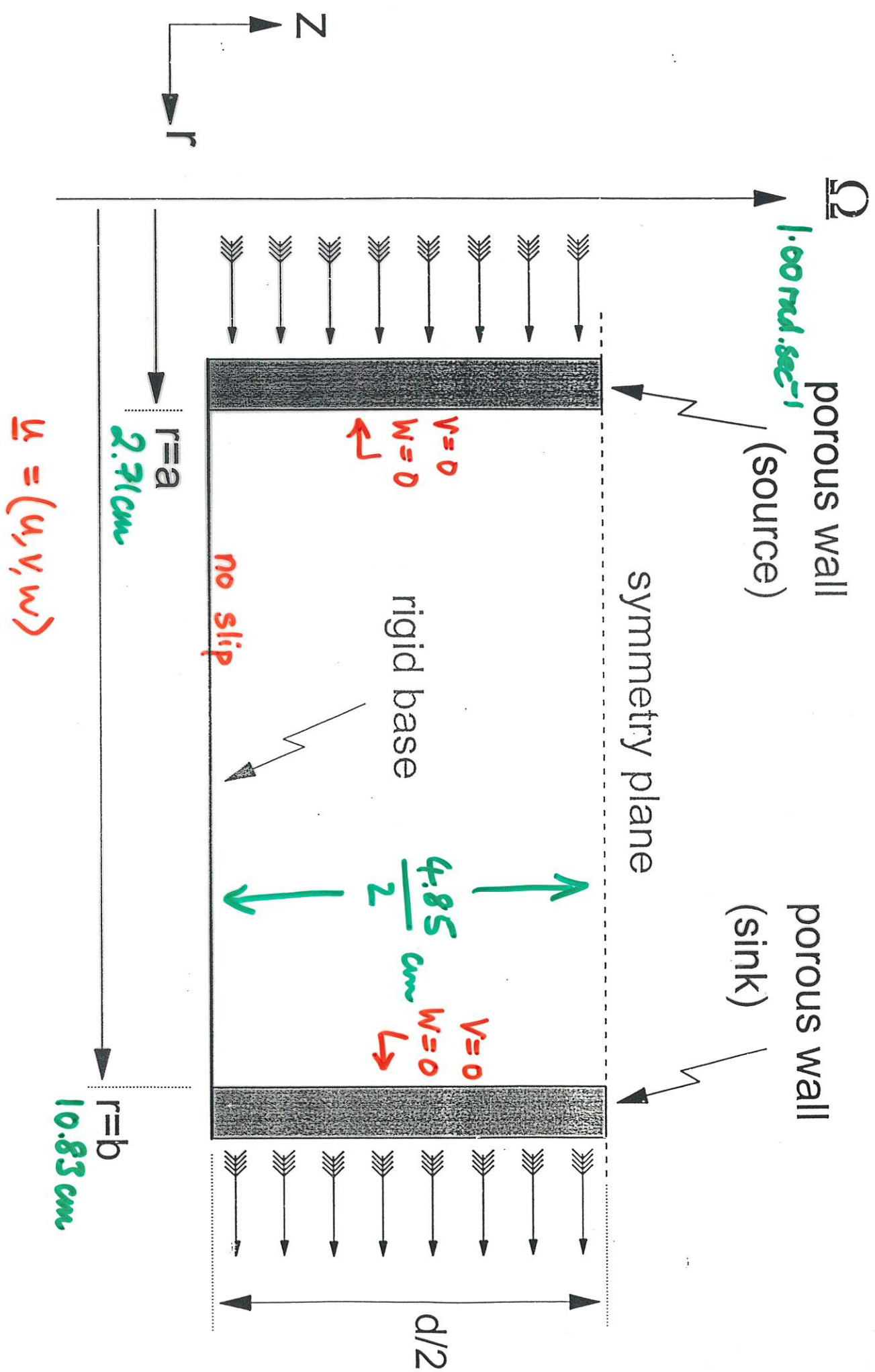
- Rigid, impermeable lid & base.
- Rigid porous cylindrical walls.
- Uniform rotation about central axis of symmetry.
- Walls, lid & base all rotate together.

3A

THE SYSTEM

(from above)

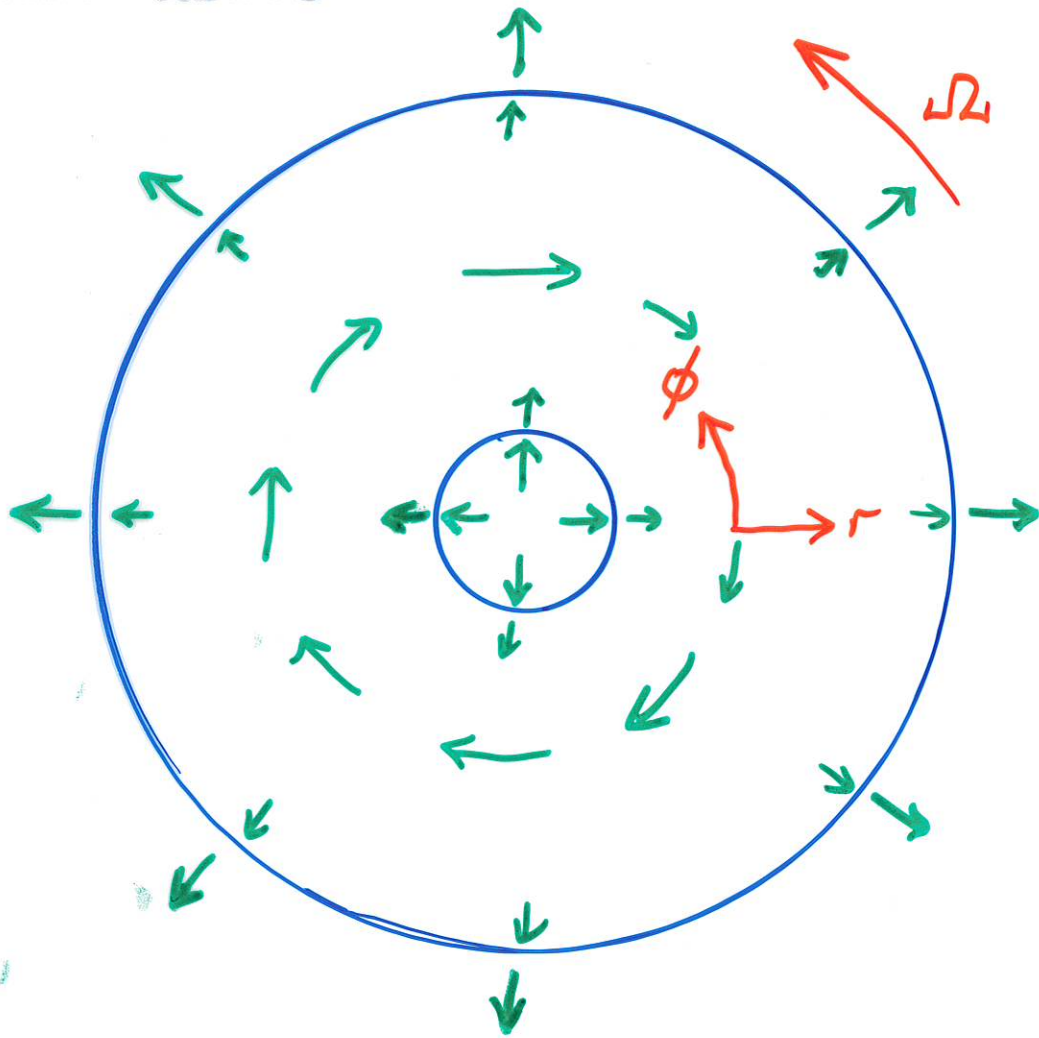




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THE FLOW :

From above

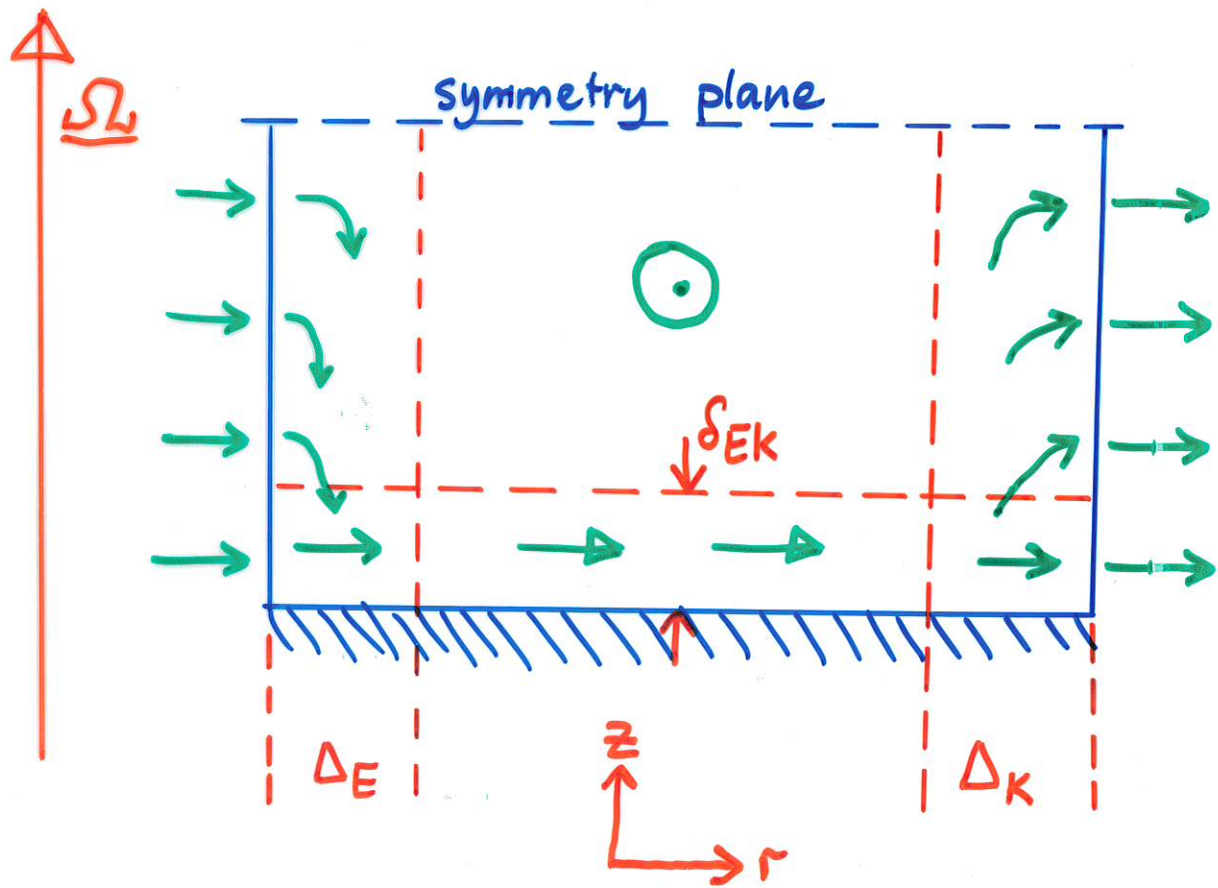


Pumped fluid feels the effect of Coriolis acceleration  $-2 \underline{\Omega} \times \underline{u}$  and turns to the right in the fluid interior.

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## THE FLOW

In  $r$ - $z$  plane.



- Azimuthal flow in fluid interior
- Radial flow in endwall boundary layers - Ekman, thickness  $\delta_{EK} = 3\sqrt{\frac{\nu}{\Omega}}$
- Vertical flow in side-wall boundary layers:
  - SOURCE layer thickness  $\Delta_E$
  - SINK layer thickness  $\Delta_K$

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## PREVIOUS WORK.

- Experiments (Hide 1968, Bennetts & Jackson 1974)
  - Azimuthal flow profiles
  - Estimates of  $\Delta E$  &  $\Delta K$
- Theory (Hide 1968, Barcilon 1970, Bennetts & Hocking 1973)
  - Predictions of  $\Delta E$  &  $\Delta K$
- A single computational study (Bennetts & Jackson 1974)
  - limited number of cases
  - Flow profiles
  - Spot-checks relative  $\Delta E$  &  $\Delta K$



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## SIDE-WALL BOUNDARY LAYERS.

- Typically
- Fix rotation rate,  $\Omega$
  - Vary total flow through system  $Q$ .

### Dimensionless Numbers

Rossby  $N^\circ = \frac{V}{\Omega d}$  , Ekman  $N^\circ = \frac{\nu}{\Omega d^2}$

### Linear Regime

$Ro \ll Ek^{1/4} \ll 1$  "Low"  $Q$ .

$\Delta_E, \Delta_K \sim \Delta_{Stewartson} = d Ek^{1/4}$

Can neglect inertial effects.

### Non-linear Regime

$1 \gg Ro \gg Ek^{1/4}$  "High"  $Q$ .

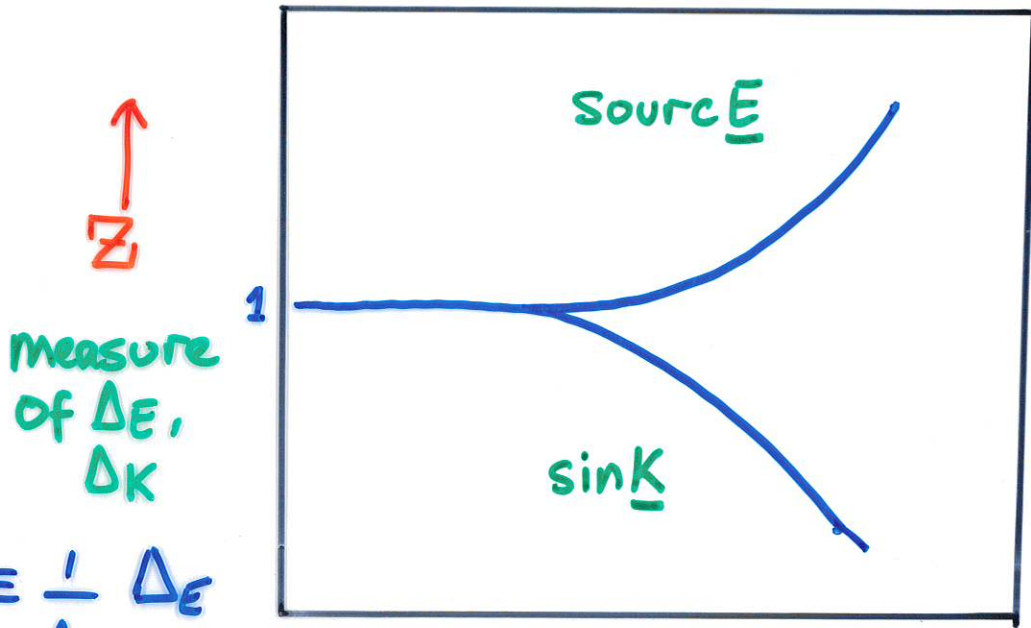
$\Delta_E \rightarrow O(Ro)$  Thickens

$\Delta_K \rightarrow O(Ek^{1/2} Ro^{-1})$  Thins.

Azimuthal flow profile "skews"

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# SOURCE & SINK LAYERS.

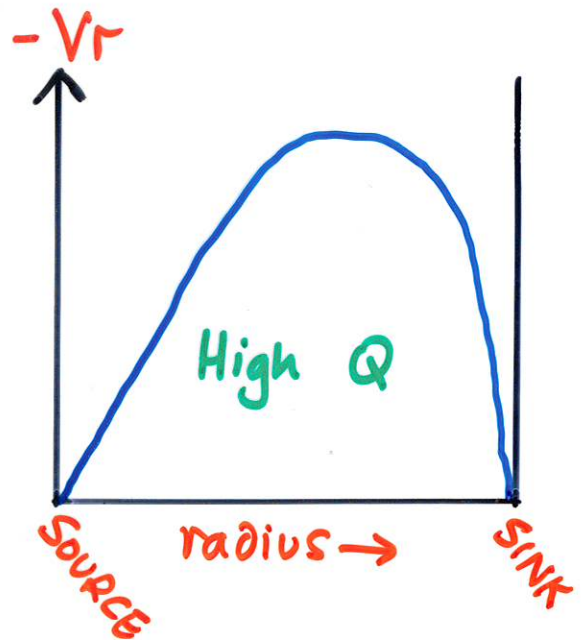
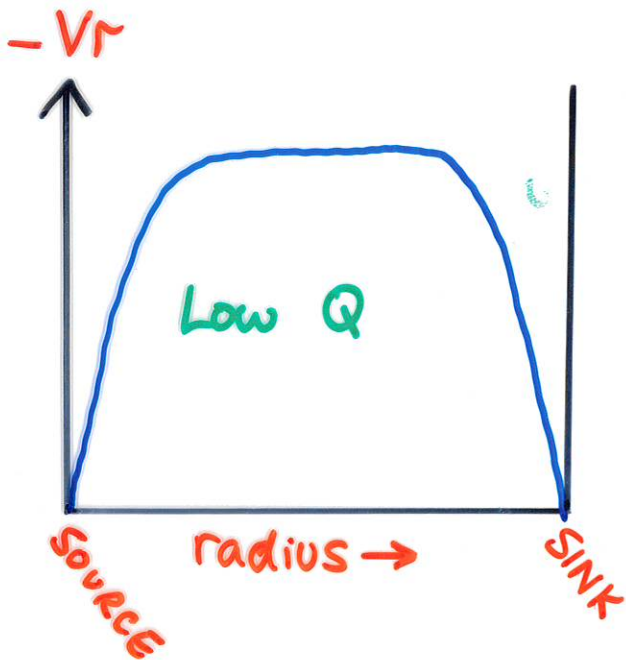


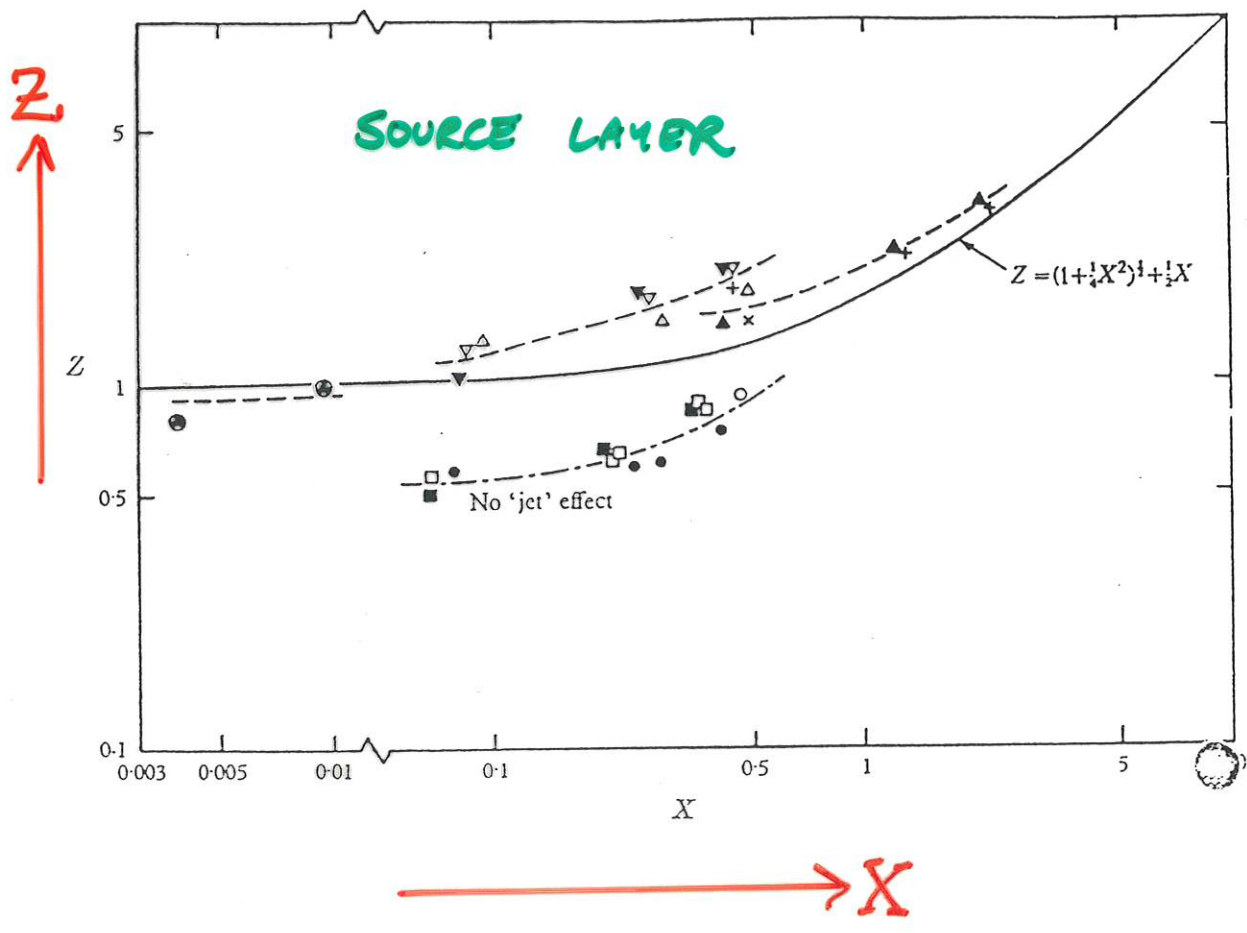
MEASURE OF  $\Delta E$ ,  $\Delta K$

$$z_E \equiv \frac{1}{\Delta_{stew}} \Delta E$$

MEASURE OF Q

$$X \equiv \frac{R_0 L}{\Delta_{stew}} \sim R_0 E k^{-1/4}$$





Hide (1968)

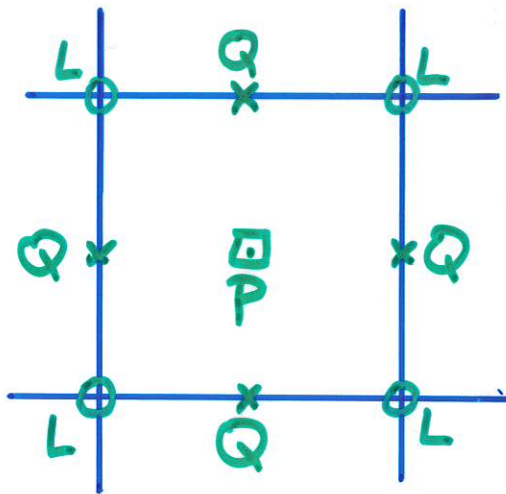
Experimental determinations of source layer thickness.

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## FEAT.

- Finite - Element
- Quadratic Interpolation
- Newton - Raphson solver (implicit)

## MESH.



Linear interpolation - L-nodes

Quadratic interpolation - L + Q nodes

$u, T, p, v$  etc. on L + Q nodes

Pressure on L + P nodes to improve representation of continuity equation.

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## FEAT (cont.)

### Finite Element :

Galerkin Finite-Elements so:-

- no built-in upwinding
- centred-type scheme.

### Newton-Raphson Solver.

- Uses analytic expressions for derivatives

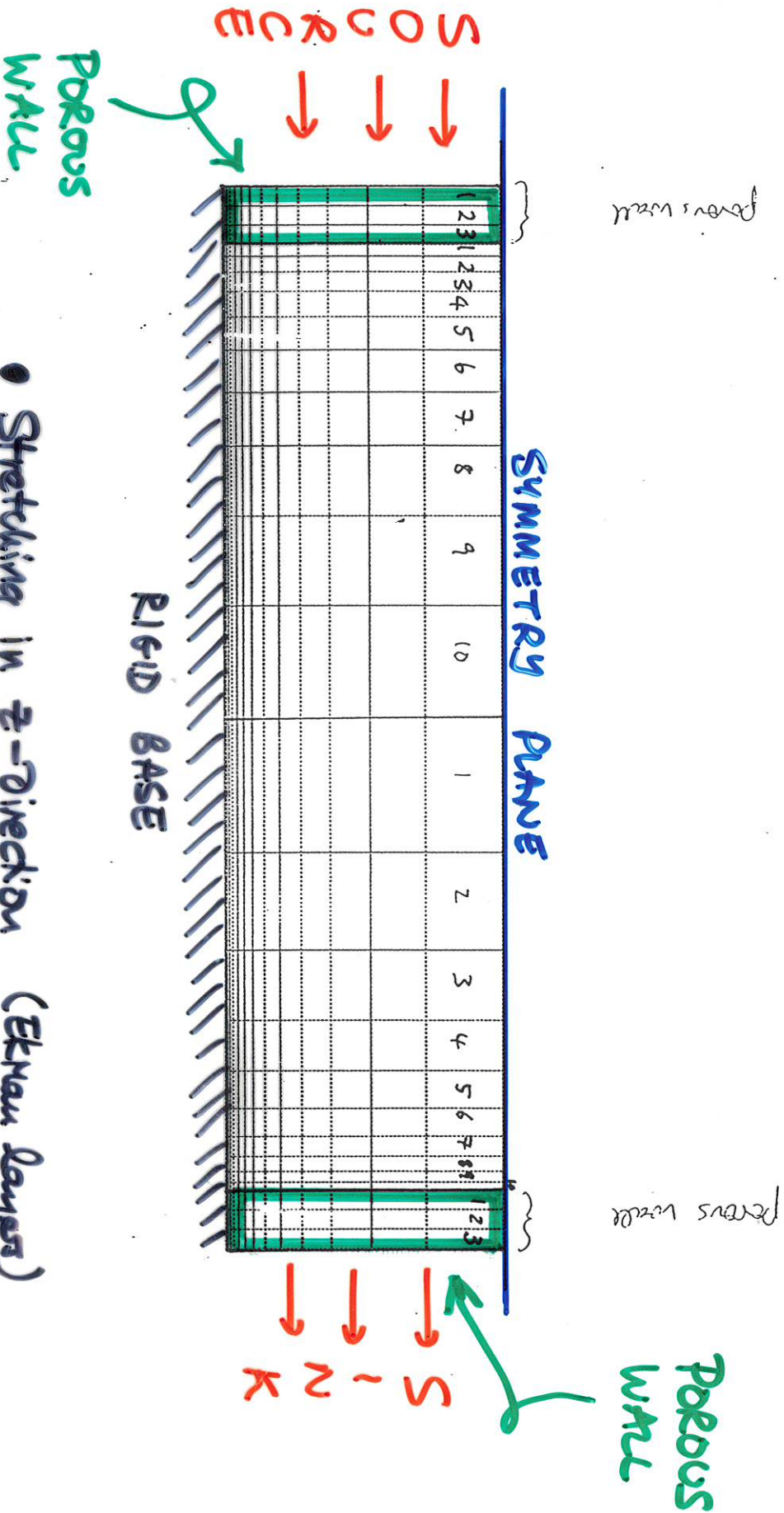
So:-

$$\text{error}_{N+1} = (\text{error}_N)^2$$

Giving faster convergence than if derivatives were calculated numerically.

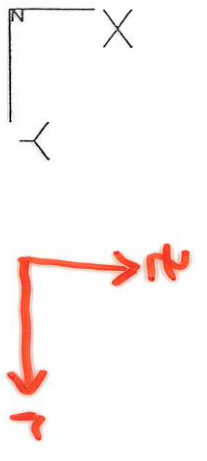
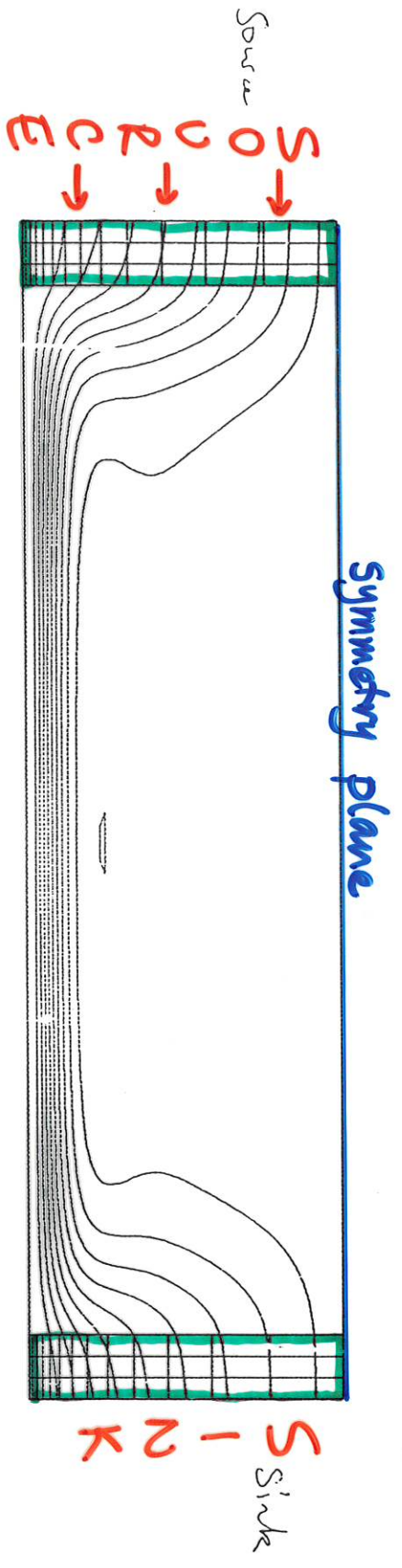
- Each iteration step matrix equation solved directly by Gaussian Elimination.

# TYPICAL MESH.



- Stretching in z-direction (Ekman layers)
- Stretching towards source & sink

# STREAM LINES



$$Q = 0.25 \text{ m}^3/\text{sec}$$

$$U_{\text{vel}} = 1.0 \times 10^{-5} \text{ m/sec}$$

$$Q = 6.1 \times 10^{-2} \text{ cm}^3/\text{sec}$$

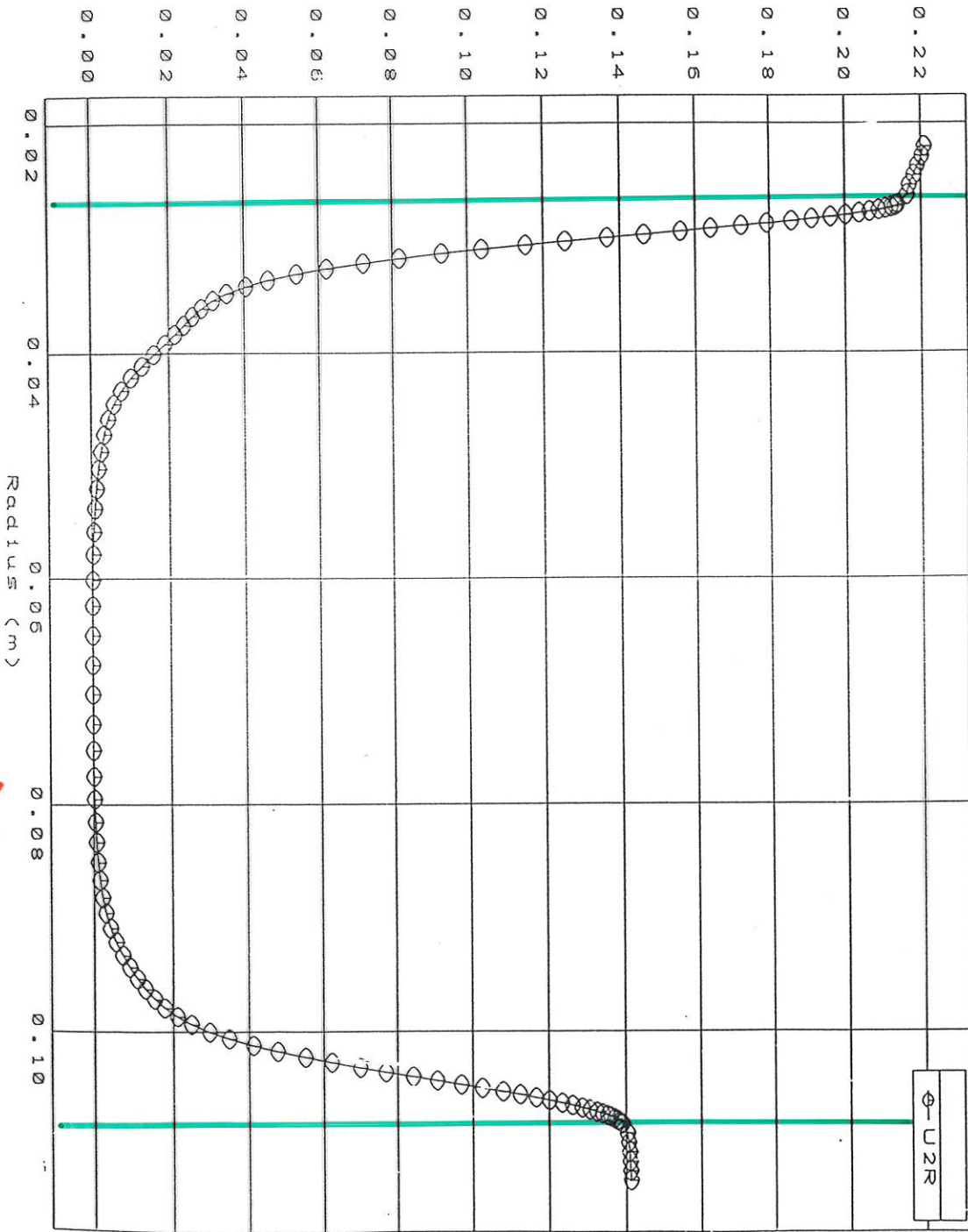
# Radial Velocity Component (x Radius) - Symmetry Plane

(Low Q)

$$\omega = 1 \text{ rev/s}$$
$$U_{\text{ref}} = 1 \times 10^{-5} \text{ m/s}$$

Radial Velocity \* Radius (m<sup>2</sup>/s) (x10E-06)

$U \times r$

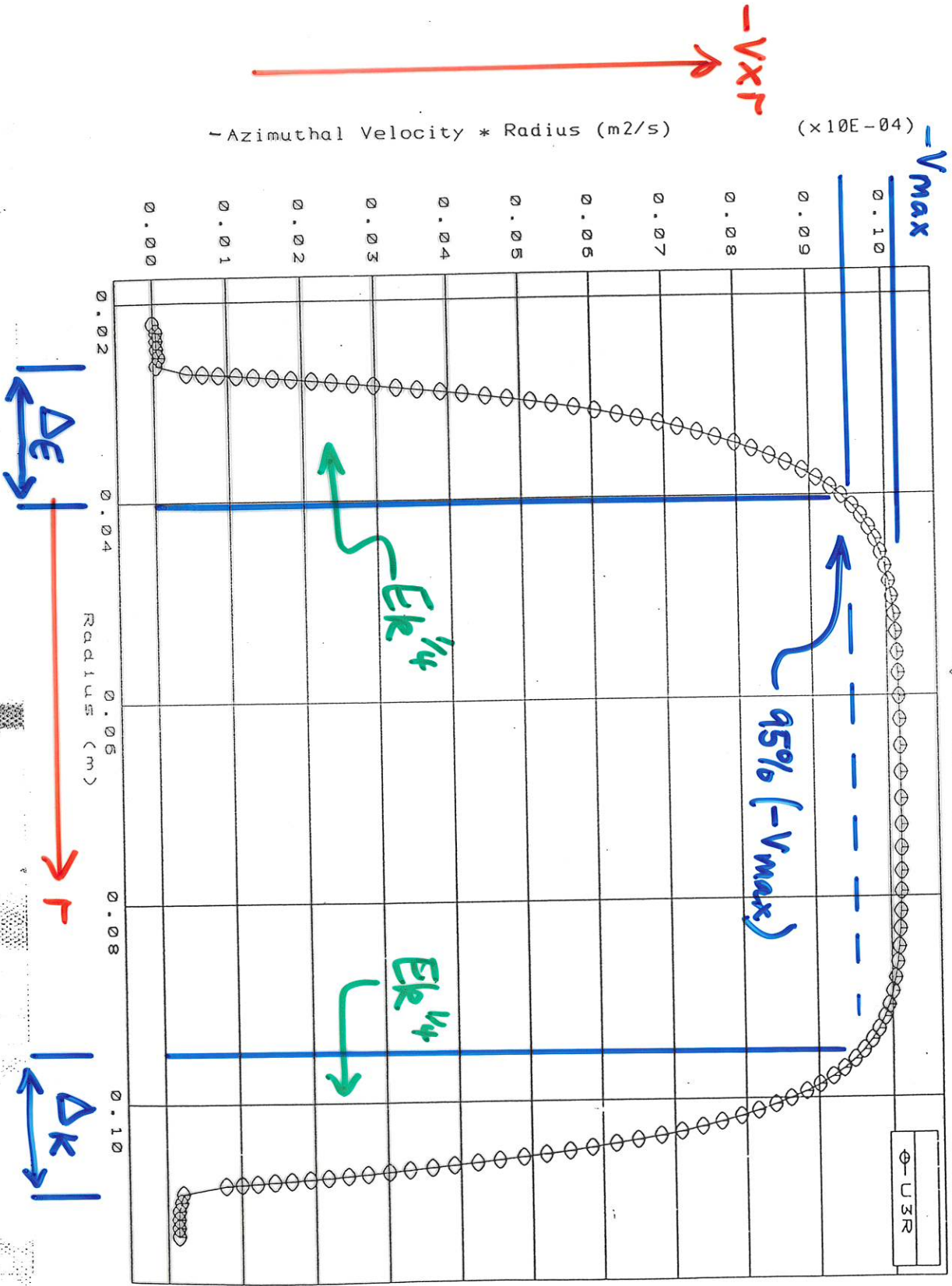


$r$



(-r) Azimuthal Velocity (x radius) - symmetry plane (low Q)

$\Omega = 111/s$   
 $u_{ref} = 1 \text{ m/s}$



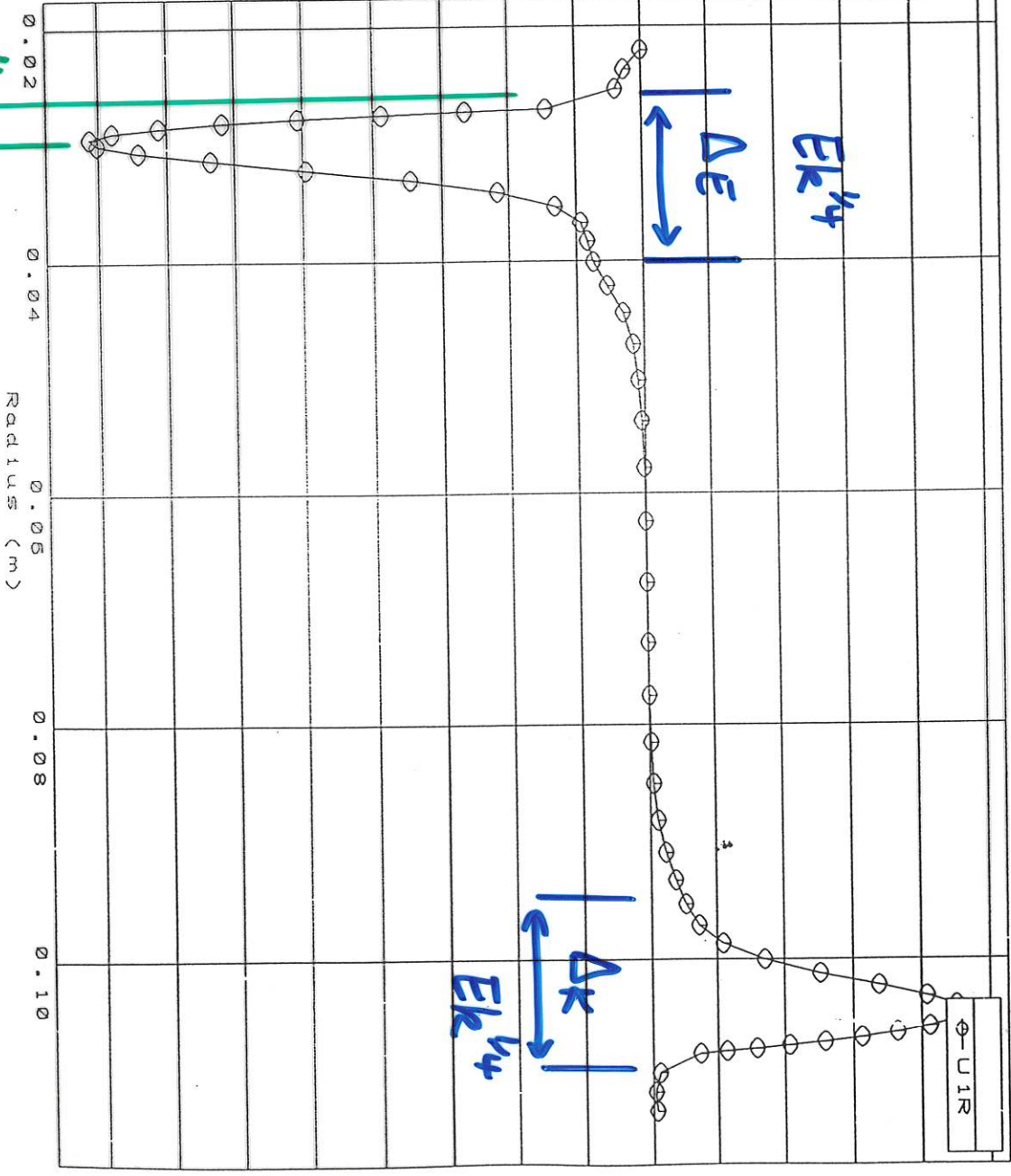
Vertical Velocity (x Radius) - Quarter Height. (Low Q)

$\Omega = 1 \text{ r/s}$   
 $U_{ref} = 1 \times 10^{-5} \text{ m/s}$

$\downarrow EK^3$

Vertical Velocity \* Radius (m<sup>2</sup>/s) (x10E-06)

0.25  
 0.20  
 0.15  
 0.10  
 0.05  
 0.00  
 -0.05  
 -0.10  
 -0.15  
 -0.20  
 -0.25  
 -0.30  
 -0.35  
 -0.40

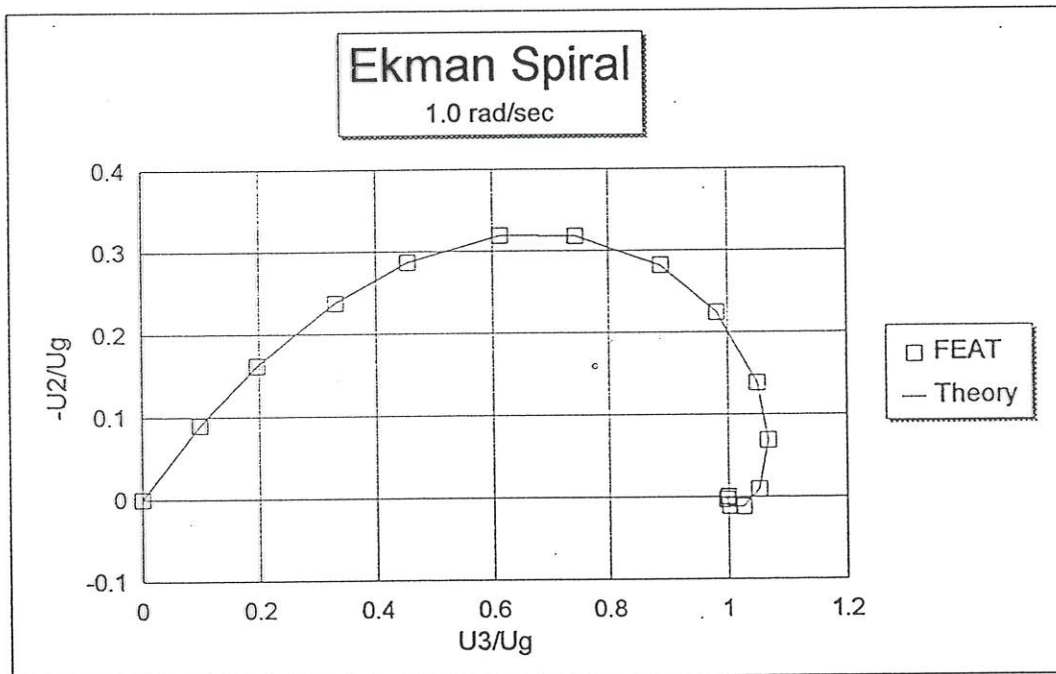


$\rightarrow WxT$

$\rightarrow EK^3$

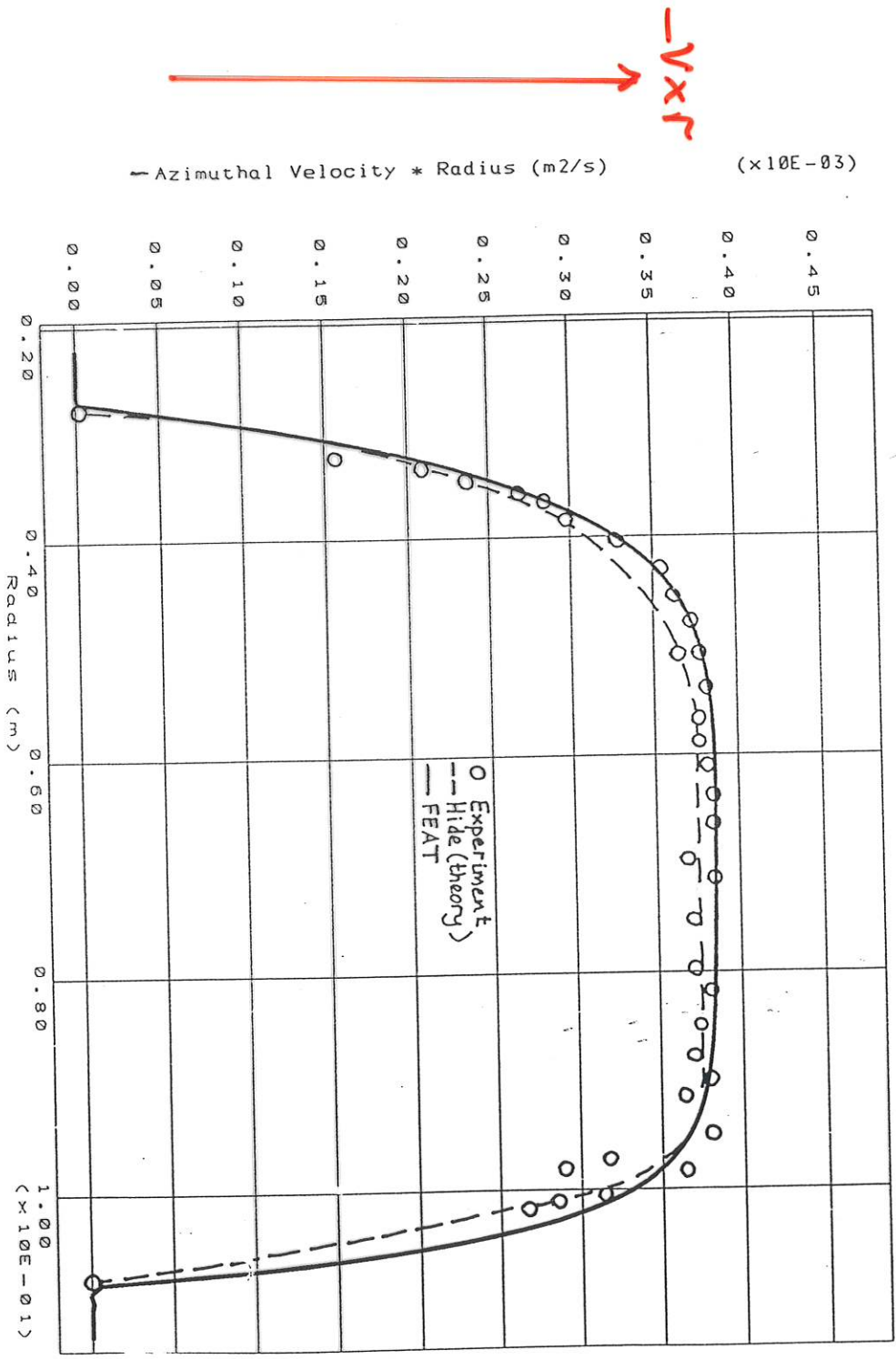
$\rightarrow r$

$\phi - UIR$



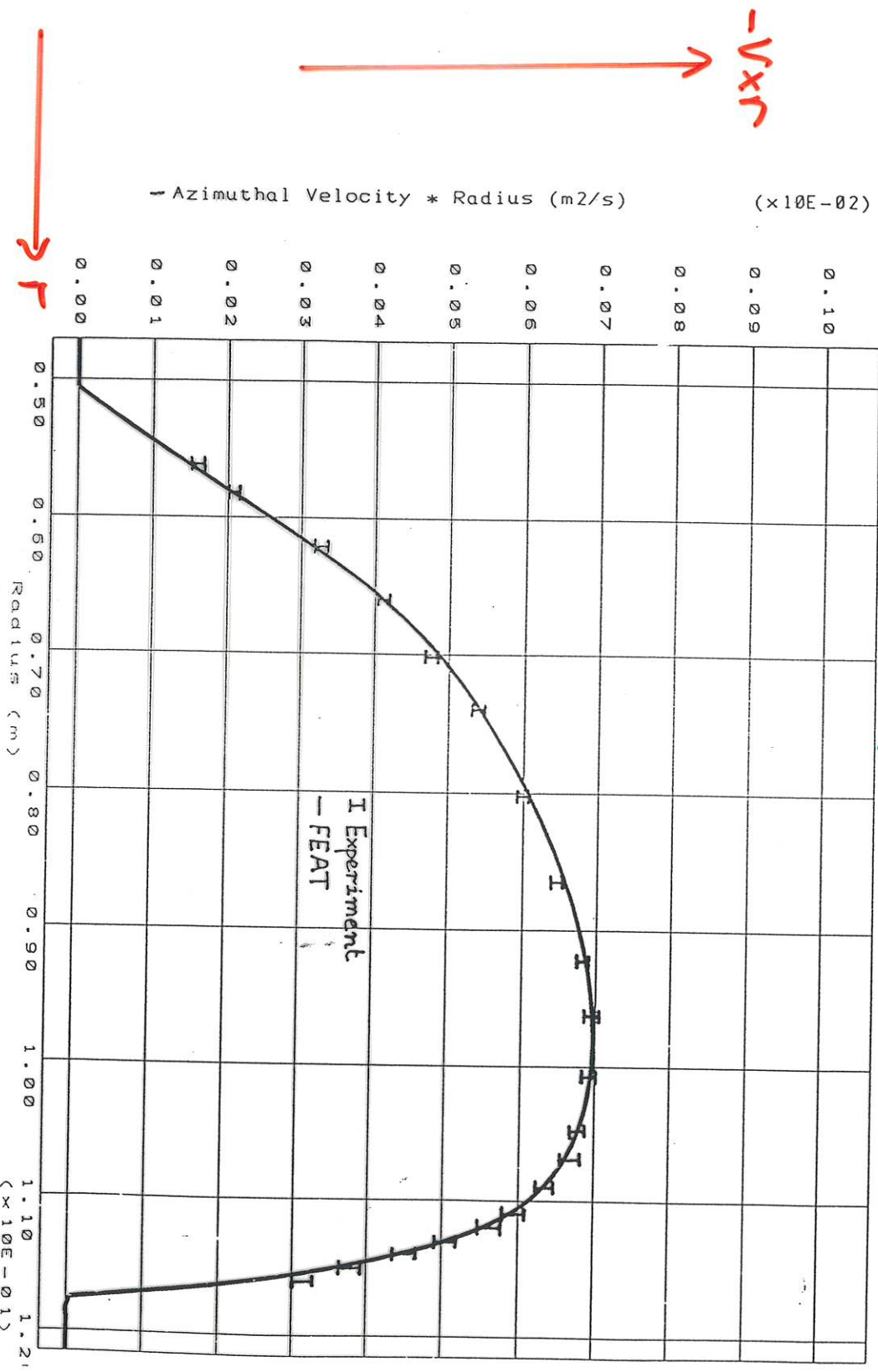
Hide (1968) — measured ○  
 — theory — — —  
 FEAT —

Low Q



Bennetts & Jackson (1974), measured I  
 FEAT

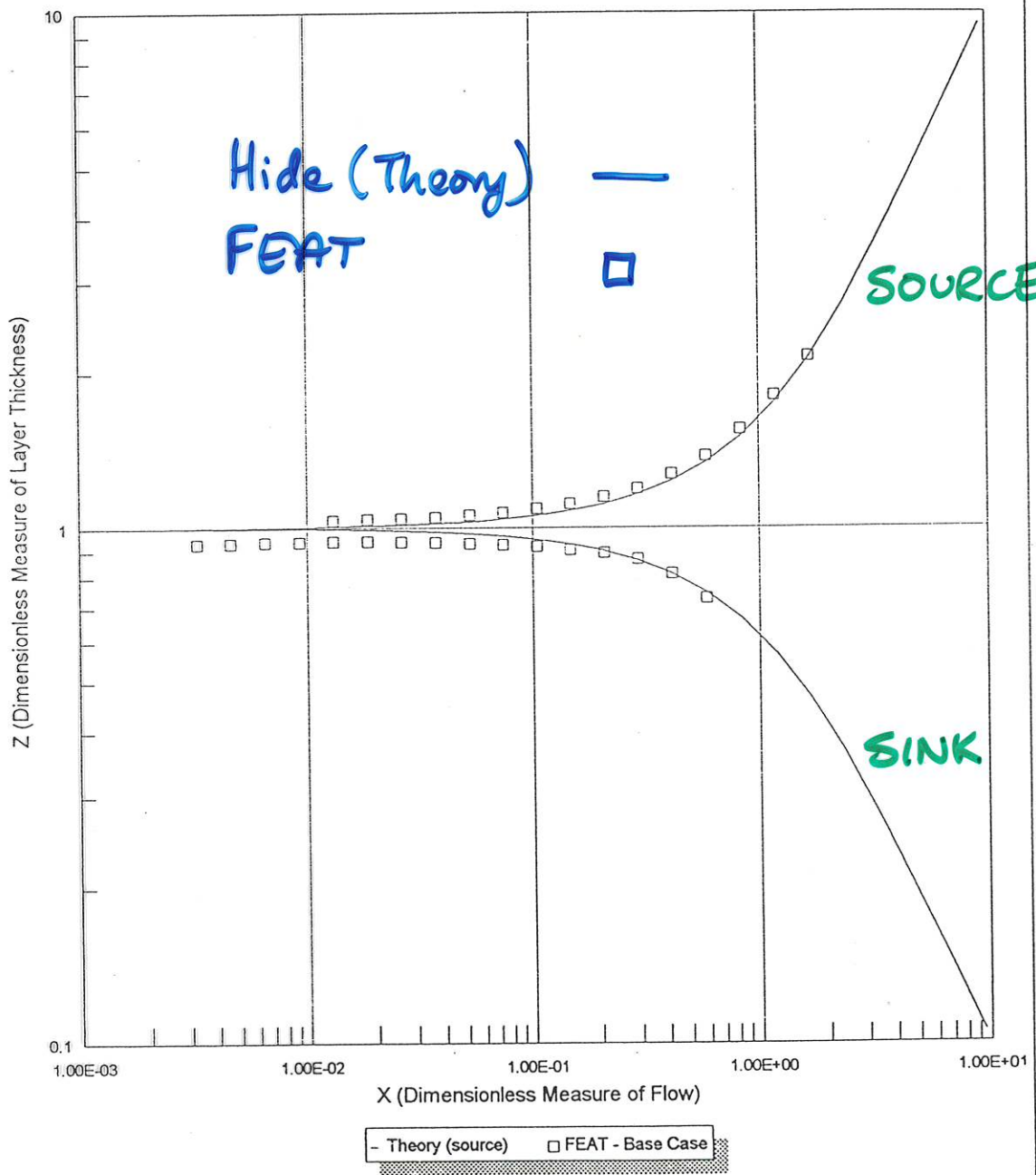
High Q



$\Omega = 3 \times 10^5 \text{ s}^{-1}$   
 $E_k = 8 \times 10^4$

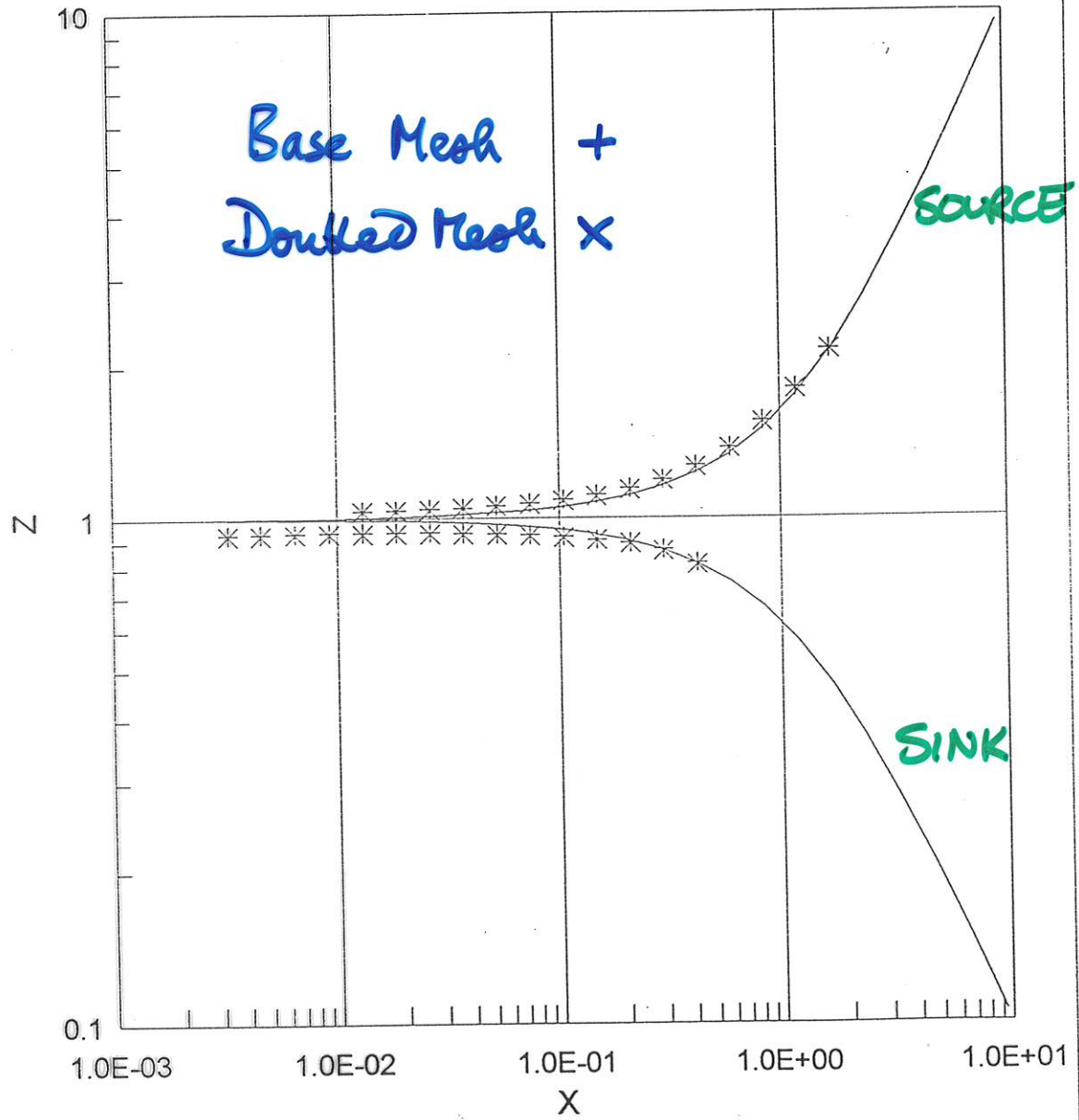
### Source & Sink Boundary Layer Thickness

Following Hide (1968)



X

### Effect of Mesh Refinement

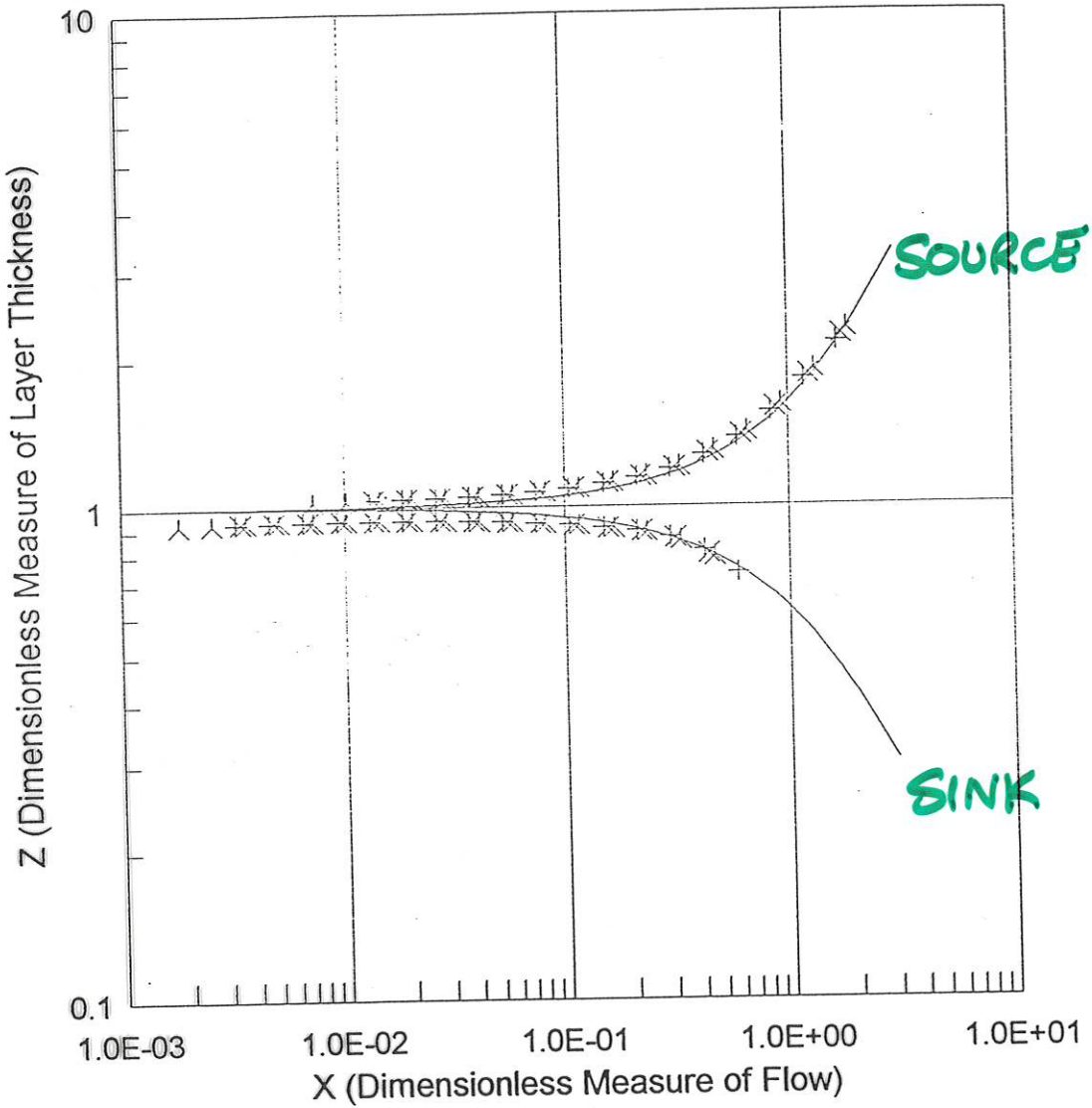


- Theory + [3x15, 25x15, 25x15, 3x15]  
x [6x30, 50x30, 50x30, 6x30]

X

# Effect of Inlet Velocity Profile

Z



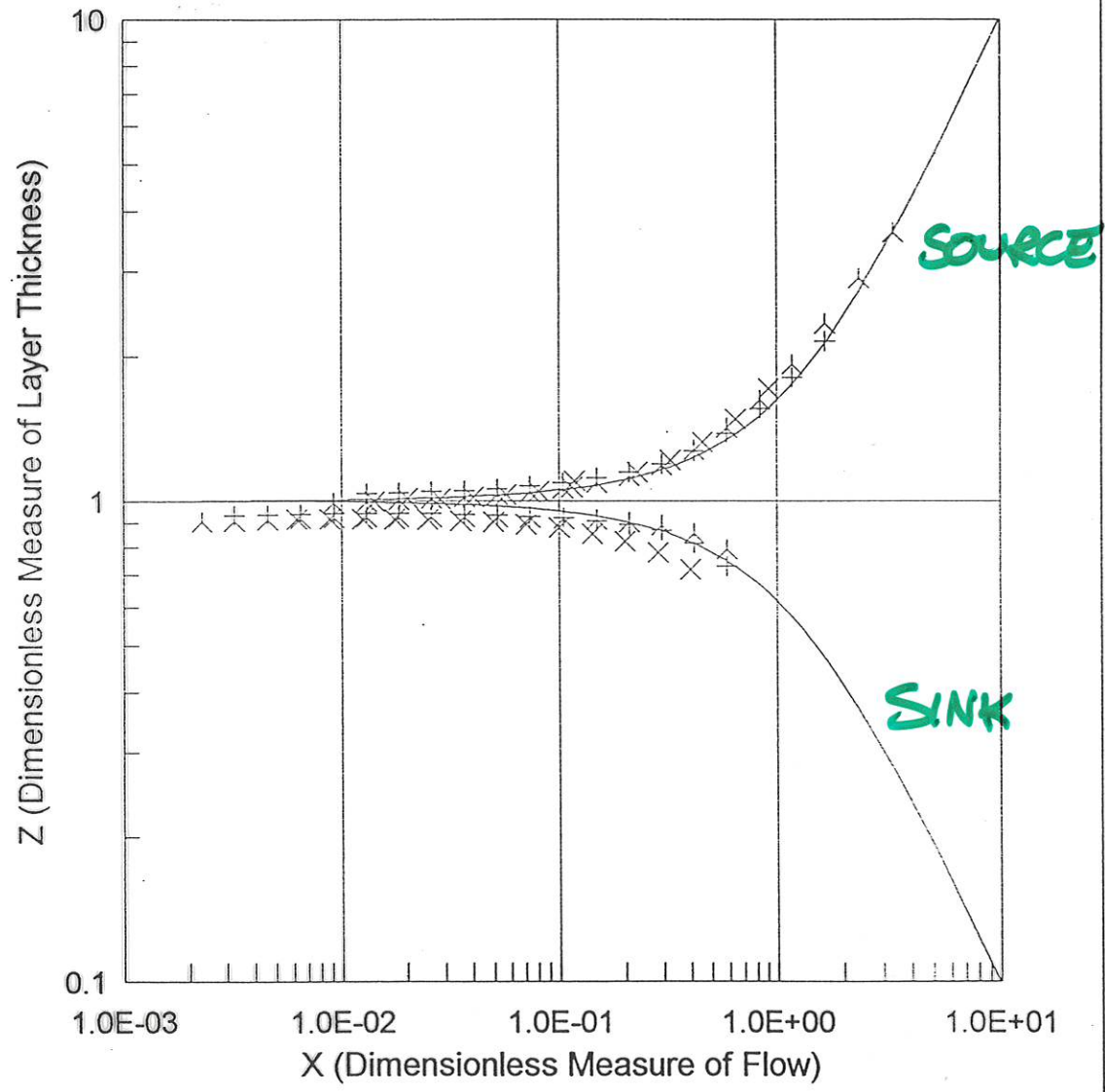
- Theory                      × FEAT n=18  
 + FEAT Base Case          △ FEAT n=1  
    **n=9**

X

$$u(z) = u_{ref} \left( 1 + \frac{2z}{d} \right)^{1/n}$$



# Effect of Geometry

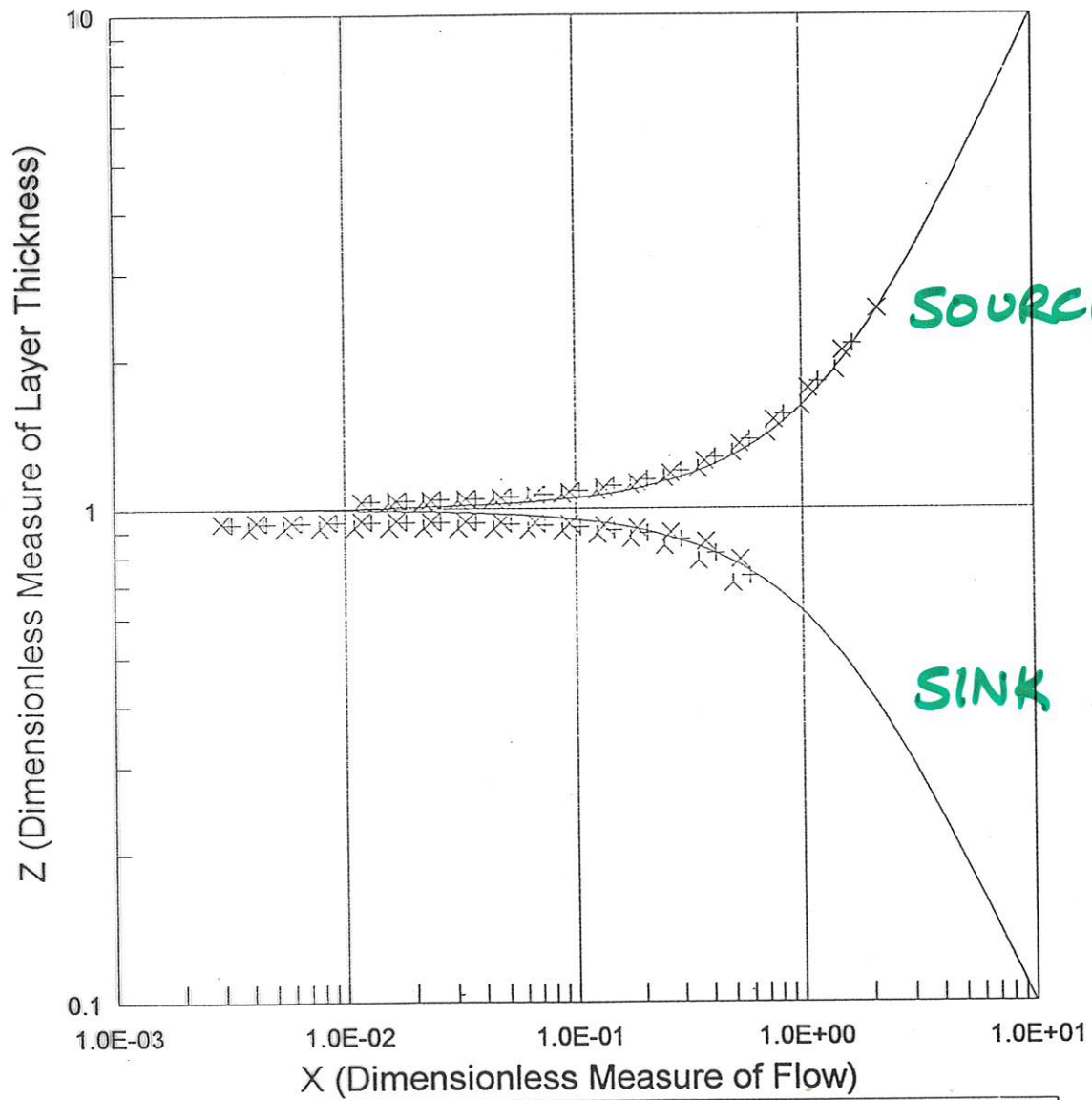


- Theory                    x FEAT 3/4 Width  
+ FEAT Base Case       ^ FEAT Half Depth

Z

X

### Effect of Rotation Rate



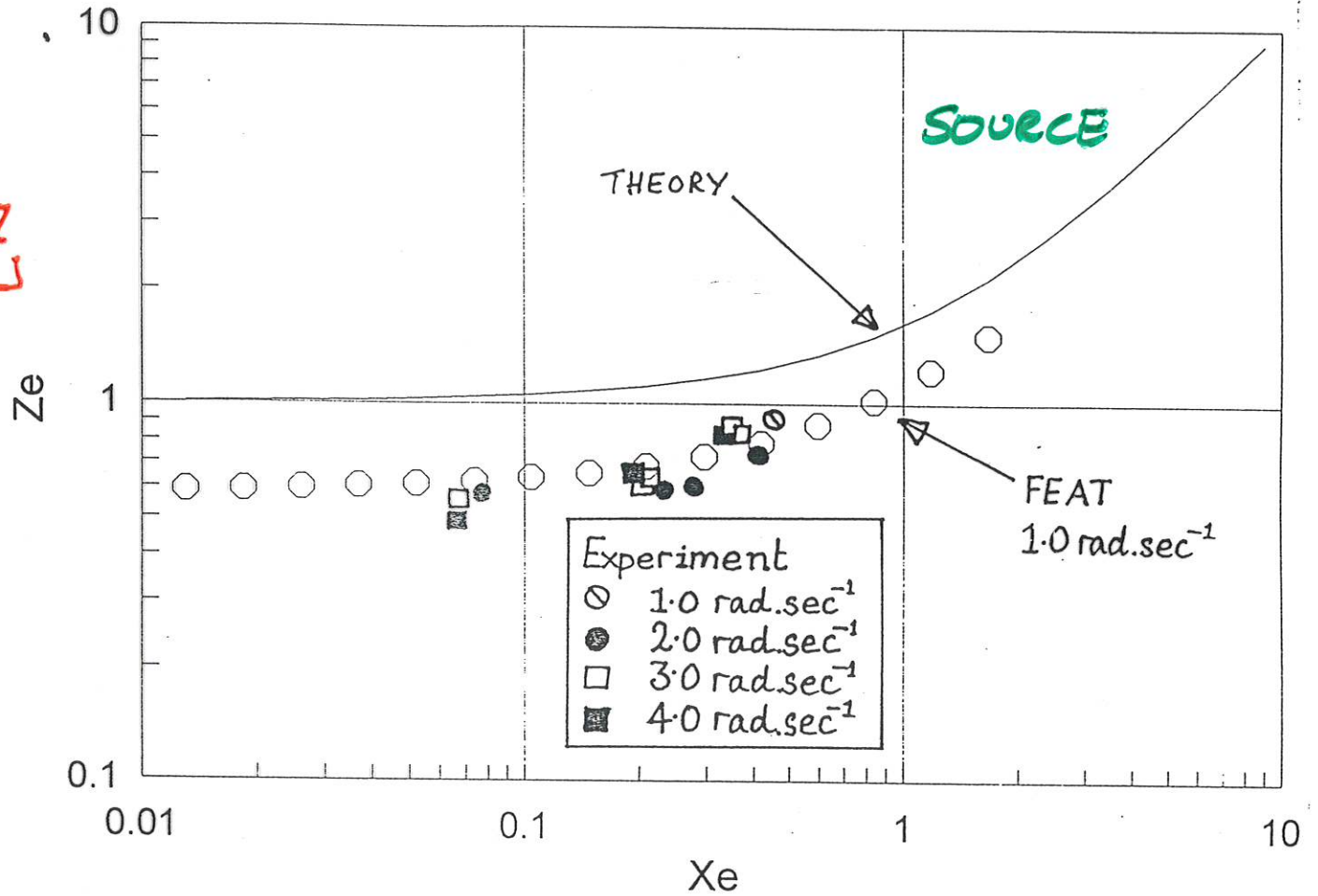
Z

- Theory (source)    x FEAT 1.5 rad/sec  
+ FEAT 1.0 rad/sec    ^ FEAT 0.5 rad/sec

X

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Σ



X

FEAT 82% boundary-layer thickness.

Source Layer			Sink Layer		
$X_E$	$Z_E$		$X_K$	$Z_K$	
	Hide (1968)	FEAT		Hide (1968)	FEAT
$1.304 \times 10^{-2}$	1.007	1.031	$3.262 \times 10^{-3}$	0.998	0.926
$1.844 \times 10^{-2}$	1.009	1.036	$4.613 \times 10^{-3}$	0.998	0.929
$2.607 \times 10^{-2}$	1.013	1.040	$6.524 \times 10^{-3}$	0.997	0.931
$3.687 \times 10^{-2}$	1.019	1.046	$9.226 \times 10^{-3}$	0.995	0.932
$5.214 \times 10^{-2}$	1.026	1.055	$1.305 \times 10^{-2}$	0.993	0.933
$7.374 \times 10^{-2}$	1.038	1.066	$1.845 \times 10^{-2}$	0.991	0.933
$1.043 \times 10^{-1}$	1.054	1.081	$2.610 \times 10^{-2}$	0.987	0.932
$1.475 \times 10^{-1}$	1.076	1.104	$3.691 \times 10^{-2}$	0.982	0.930
$2.086 \times 10^{-1}$	1.110	1.136	$5.219 \times 10^{-2}$	0.974	0.927
$2.950 \times 10^{-1}$	1.158	1.186	$7.381 \times 10^{-2}$	0.964	0.922
$4.172 \times 10^{-1}$	1.230	1.260	$1.044 \times 10^{-1}$	0.949	0.915
$5.899 \times 10^{-1}$	1.338	1.374	$1.476 \times 10^{-1}$	0.929	0.905
$8.343 \times 10^{-1}$	1.501	1.546	$2.088 \times 10^{-1}$	0.901	0.890
1.180	1.751	1.798	$2.952 \times 10^{-1}$	0.863	0.861
1.669	2.137	2.148	$4.175 \times 10^{-1}$	0.813	0.808

FEAT Solution for Z Compared with Hide's (1968) Approximate Theory.

When source on inner cylinder

FEAT source layer slightly thicker than Hide's theory at low Q.

Sink layer slightly thinner.

Probably due to curvature (reverse flow direction, put source on outer cylinder to test)

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## SUMMARY.

- FEAT Validated against isothermal rotating fluid flow.
  - azimuthal flow profiles
  - Ekman b. layers
  - Source & Sink b. layers.
- Shown systematic agreement with theory for  $\Delta_E$  &  $\Delta_K$  over range of  $Ro$ ,  $Ek$  considered.
- Shown that Hide's result is generally independent of:
  - inlet velocity profile
  - system depth & width changes
  - rotation rate

NEXT -

Reverse flow direction  
Increase depth.